Towards model-based approaches for musical instruments making: validation of the model of a Spanish guitar soundboard and characterization features proposal

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Abstract

Nowadays, the virtual prototyping method is widely used for industrial applications and can lead to a powerful tool for musical instruments making and conservation. Nevertheless, physics-based models of musical instruments are barely developed for this purpose and the confrontation between model predictions and experiments have been the focus of very few researches. The objective of this paper is to highlight the predictive capability of physics-based models in dynamic domain, even in presence of variable by nature material and climatic conditions. For this purpose, a finite element model of the soundboard of a Spanish guitar is developed for model validation purposes. The simulated modal bases are compared with experimental ones from a previous study. Screening and stochastic analyses are performed to rank which are, among material and climatic parameters, the most influential ones on the dynamics of guitar soundboard. Moreover, uncertainties are taken into account to evaluate the dispersion of the response for a given design, and simulations are validated facing experimental data. It is shown that specific elastic parameters of the wood (in longitudinal and radial directions and longitudinal-radial plane) of the top plate are mainly

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influential with regard to the dynamics of the soundboard, and the relative humidity changes have a non negligible impact. Moreover, test-model correlations have shown that a nominal model with average material parameters is able to predict the dynamical behaviour of a real braced soundboard with an average error on the first eight eigenfrequencies lower than 4%. In addition, when uncertainties are taken into account, the model is able to predict every experimental data. Finally, dynamic features like CFDAC and Fuzzy-FRFs are proposed in an innovative way in this application domain.

Keywords: Model validation, Uncertainty quantification, Modal analysis, Screening analysis, Musical acoustics, Guitar soundboard

Higlights

- Model validation of musical instrument part even in presence of strong uncertainties
- Fuzzy-FRF method illustrated to musical acoustics domain
- Screening analysis of influential material and climatic parameters on guitar soundboard dynamics for fixed geometry
- Similar impact of $\frac{E_L}{\rho}$, $\frac{E_R}{\rho}$, $\frac{G_{LR}}{\rho}$ and relative humidity on eigenfrequencies of guitar soundboard in free conditions
- Predominance of $\frac{E_L}{\rho}$, followed by $\frac{E_R}{\rho}$, $\frac{G_{LR}}{\rho}$ on eigenmode shapes

1. Introduction

- The guitar, electric or acoustic, is one of the most popular musical instru-
- ment all over the world. It has been used in almost every musical style since
- 4 the XX^{th} century and is often the first choice for musical learning. The gui-
- 5 tar clusters a wide range of prices and qualities. One main family is considered
- 6 here, the acoustic guitars. The acoustic guitars transfer the vibratory energy of
- 7 the strings partly into an acoustic energy through their sounding box and the
- 8 radiation of their flat parts, especially soundboards.

It is usually considered that the soundboard affects mainly the acoustic behaviour of the guitars [1], [2]. The shape and thickness of the soundboard, as well as the characteristics of the braces (number, orientation, shapes) affect the mechanical behaviour of the guitar in both the static and dynamic domain. In the static domain, the top plate has to sustain the strings tension, and the mecha-13 nical characteristics of both plate and braces affect the stiffness of the system. 14 In the dynamical domain, the eigenmodes of the soundboard are also strongly linked with the mechanical behaviour of the parts. These different design and material choices, that depend on the instrument maker, impact the resonance 17 modes and acoustical features of the instrument. However, these parameters are barely quantified and studied from an objective point of view, since the 19 construction process is based mainly on tradition and empiricism. Generally, mechanically based works study acoustic guitars by experimental, analytical and numerical means, mainly focusing on modal parameters to esta-22

23 24 blish comparisons.

The experimental means have been used for decades to observe the resonance 25 modes of the guitar soundboards, alone, or when coupled with the sides and the 26 remaining parts. As an example, it has been used for the objective characte-27 risation of different guitar families, related to their soundboard braces pattern [3]. The experimental means are also useful for different purposes, such as the 29 study of the global dynamics of the guitar and the radiated sound produced, dealing with a macro response of the instrument [4, 5]. This type of response contains a high amount of information, and it becomes hard, considering the 32 total coupling of all the elements, to attribute the role of each part of the guitar 33 (and the properties of its components) with respect to the observed response. 34 Specific numerical models related to guitar have been developped since 70's [6], up to a detailed model in [7]. More recently, the evolution of the computational power enabled increasingly sophisticated models such as the complete 37 channel of the production of sound, from the plucked guitar string to the radited sound [8]. One of the most complex model, mixing complete structure and fluid structure interactions has been developed in [9, 10] and enabled the computation of the radiated sound around the instrument. In [11], a detailed vibroacoustic model of a Portuguese guitar has been compared, in a deterministic way, with a real instrument. Usually, models of soundboards include braces, that have also been studied separately in [12]. In addition to the computation of the modal basis of guitar soundboards, the models have been used to compute the bridge admittance of the guitar, in a similar way than the one considered for violins, and thus provided results that could be compared with easily measurable features of real guitars [13].

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So, historically, models were used as *a posteriori*, but, nowadays, increasingly for the prediction of complex structures. Physics-based modelling is used to predict the mechanical behaviour of complex virtual systems in the first steps of prototyping, and to quantify the variability of its behaviour, submitted to numerous unavoidable sources of uncertainties. Nevertheless, as the utilisation of the models concern more sophisticated structures, the models are still often unable to correctly predict their behaviour.

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Thus, in order to avoid issues inherent to model prediction erros, the verifica-58 tion and validation (V&V) process has been developed to assess the viability of a 59 model, and its framework is detailed in [14] and the book [15]. The V&V method 60 needs a large number of simulations to predict as much cases as possible, that are parts of the uncertainty domains. Its application on the vibratory behaviour of structures has been performed in [16, 17, 18]. The validation process aims at 63 ensuring the reliability of the model when predicting the behaviour of a system. The predictions of the model (that can consist of numerous features) are com-65 pared to experimental data. The model is validated when the closeness between numerical and experimental results is below a tolerance level. The experiments make perfect sense when the material used exhibits a high variability, like the wood. Therefore, a higher number of experiments leads to an enhancement of the model reliability, through the validation criterion.

This paper aims at highlighting in an innovative way for this application 72 domain the potential of numerical models and their relevance in regard with musical acoustic applications. Indeed, in the framework of the utilisation of numerical models for the musical instrument making, the reliability of the models have to be at first time assessed and quantified, which is the main objective 76 of this paper. In this study, a model validation of a musical instrument part is proposed. Experiments have been carried out on five similar Spanish guitar soundboards at 15 different steps [19, 20]. Usually these steps performed 79 by instrument makers are led by empiricism and traditions, and barely quantified to highlight objective assessments. In the case of wooden parts of musical 81 instruments, the variability of such material is high and inevitable. Thus, this aleatory uncertainty is taken into account with stochastic approaches. Sensitivity analysis have to be performed to evaluate the relative influence of material parameters. Screened material parameters will be implemented with uncertainty model. The climatic conditions are also a source of uncertainty that is modelled with probabilistic approaches. 87

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In the next section, the model development and methods used, as well as the parameters of the material behaviour's law are given. The results give dynamic data, sensitivity analysis results and dispersion of the computed eigenfrequencies with comparison with experimental ones. The conclusion give the main results and advances of this paper.

94 2. Model and methods

5 2.1. Experimental modal analysis of Spanish guitar soundboard

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The modal analysis of the guitar soundboard is the same as the one performed in [19], the material and methods is based on this paper. The numerical

model and experimental data used correspond to the construction stage № 15 of the paper [19], where all the braces and reinforcements are glued on the sound-101 board, and the rosette has been glued in its cavity. So, this stage corresponds to 102 the state of the soundboard before being glued on the sides of the guitar. The five 103 soundboards tested are labelled Sb_{01} , Sb_{02} , Sb_{03} , Sb_{04} , Sb_{05} , and their masses 104 are equal to 164, 166, 154, 178 and 170 g, respectively. The average mass of 105 the soundboards is equal to 166.4 g. Only the out-of-plane motion of the sound-106 boards has been considered. The guitar soundboards have been tested under free 107 boundary conditions. A unidirectional accelerometer (B&K 4518-003) has been 108 used and glued near the future bridge location (at the point 69 on the figure 109 1 (a)). An impact hammer (B&K 8204) has been used to excite the structure 110 on the 99 points also displayed in the same figure. The test frequency ranges 111 between 0 and 800 Hz, and the corresponding resolution is equal to 0.25 Hz. The vibration signals were measured and recorded as time series and processed 113 into inertance FRF data. Both the applied excitation and the measured response 114 were perpendicular to the soundboard. Signals were averaged two times for each 115 measurement point. A modal analysis of the inertances has been performed to 116 evaluate the modal basis of each soundboard. The modal analysis has led to 117 the evaluation of a modal basis with eigenfrequencies and corresponding modal 118 dampings for each modes. The experimental results obtained with this study, 119 [19], for the first eight modes identified for the five soundboards are displayed 120 in the table 3.

2.2. Computer aided design and meshing of the soundboard

The numerical model has been developed using the finite element method based on a Computer Aided Design (CAD). The CAD software used is SOLIDWORKS[®].

The software used for the pre-processing is PATRAN[®] and the solver is NASTRAN[®].

The figure 1 (b) represents the CAD of the soundboard and the nomenclature.

The figure 2 (a) displays the finite element model mesh. The CAD is meshed using tetrahedral elements with quadratic interpolation. The interfaces between the soundboard and the different parts are considered as perfect and are mo-

delled with coincident nodes and equivalent faces. The finite element model contains 55000 elements and 104000 nodes and free boundary conditions are applied.

2.3. Material orientation

The material used is spruce, *Picea abies*, for all the braces, bars, patches and the soundboard. The material properties are taken from [21]. The values are given in the table 1. The parts are oriented according to the figure 2 (a). The specific elastic parameters are sampled. The density d_0 is also sampled according to variations evaluated in [21]. The temperature and relative humidity are also sampled as equi-probalistic approaches bounded between 25 and 85 % for RH and 15 to 35 °C for T.

As a second step, the moisture content is calculated from the RH and T values sampled, according to [22], given in the eq. 1:

$$MC = 10 + 0.16 \times (RH - 50) - 0.03 \times (T - 21) \tag{1}$$

The density as a function of MC, ρ_{MC} , is then calculated from the value of MC, according to [23], given in the eq. 2:

$$\rho_{MC} = \rho_0 \times (1 + 0.01 \times (MC - 10)) \tag{2}$$

The elastic constants values of E_L , E_R and G_{LR} depend on the relative humidity and, in a lesser degree, on the temperature. In order to implement this dependence, the values of the elastic properties are implemented as a function of RH and T, laws are taken from [24], given in the eq. 3:

$$E_{L_{RHT}} = E_{L_{\rho}} (1 - 0.0015 \times (RH - 50) - 0.0008 \times (T - 21))$$

$$E_{R_{RHT}} = E_{R_{\rho}} (1 - 0.005 \times (RH - 50) - 0.0025 \times (T - 21))$$

$$G_{L_{R_{RHT}}} = G_{L_{R_{\rho}}} (1 - 0.007 \times (RH - 50))$$
(3)

When a set of elastic constants as a function of density $(\frac{E_i}{\rho})$ as an example is sampled, it is multiplied by a sampled value of density ρ_i . The value of E_i is

then modified following eq. 3, with sampled values of RH and T. The sampled density is then also expressed as a function of the moisture content (eq. 2, given by 1), which gives ρ_{MC_i} . So, at the end of the sampling process, E_i , G_{ij} and ρ_{MC_i} are implemented in the numerical model, for the parts made of spruce.

Four sets of material parameters are considered dedicated to the soundboard, 155 the braces, the bridge patch and the sound-hole reinforcement respectively. The 156 material used for the parts is implemented to match the orientation of the wood 157 in the reality. The wood samples are quarter-sawn and the dimensions are small 158 enough to consider an orthotropic definition. Considering the coordinate frame 159 represented in the figure 2 (b), R corresponds to the radial direction, L to the 160 longitudinal direction, T to the tangential direction. A modal basis is computed 161 with the nominal values given in the table 1. With these values, the mass of the 162 model is estimated to 168 g and the model average mass of the real soundboards is equal to 166.4 g. 164

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5 2.4. Model-based modal analysis

A modal analysis is computed with the numerical model. Starting at 1 Hz, 167 the first 50 modes are computed, which lead to a bandwidth for the modes 168 extraction between 1 and 1000 Hz. The frequency response functions (FRF) in 169 acceleration, velocity and displacement (inertances, mobility and admittance), 170 are computed on the 99 points of the experimental test, in the out-of-plane direction, Y. The driving force value is equal to 1 N, applied in the same direction 172 on the point labelled 69 in the figure 1 (a). The computation of the FRF is made 173 on the numerical nodes close to each experimental points. In the considered 174 frequency bandwidth, relative humidity and strain levels it is hypothesised that 175 the material exhibits a linear elastic behaviour. Thus the damping is applied a posteriori as a modal damping whose value corresponds to the mean modal 177 damping measured on guitar soundboards in the considered bandwidth, with 178 $\xi = 1.15\%$. This value is taken from the experimental part and given in the 179 table 3.

2.5. Screening analysis

Sensitivity analyses are computed to evaluate the impact of the variability 182 of inputs of the model with regard to eigenfrequencies and eigenvectors outputs 183 $Y(X_i)$ matched using MAC criterion [25]. Finite difference analysis is used at first to roughly screen the material and climtic parameters as a function of their 185 impact. For this purpose, a variation δX_i equal to 1 % is applied on each input 186 parameter X_i one at a time. The sensitivity indicator is given by the eq. 4 [26]: 187

$$\phi_{i} = \frac{\frac{Y(X_{1}, \dots, X_{i} + \Delta X_{i}, \dots, X_{n}) - Y(X_{1}, \dots, X_{i}, \dots, X_{n})}{Y(X_{1}, \dots, X_{i}, \dots, X_{n})}}{\frac{\Delta X_{i}}{X_{i}}}$$
(4)

The morris screening analysis [27] is used to explore the input domain. The 188 linear and coupling effects of the X_i are evaluated. For a number of n_p para-189 meters X_i in a n_p dimensions domain Ω , the domain is sampled in l levels. The values of the domain Ω are defined in order to depict all the values that the 191 parameters can attain. The elementary effect of a parameter X_i in a sample X^j 192 of the space is given by: 193

$$E_i^j = \frac{f(X^j \pm e_i) - f(X^j)}{\pm \Delta} \tag{5}$$

With e_i a unit vector and Δ a value taken in $\left\{\frac{1}{p-1},...,1-\frac{1}{p-1}\right\}$. The n_t 194 trajectories will define the number of computations given by $(n_p+1)\times(n_t)+1$. 195 For this computation, the number of levels is equal to 6, n_t is equal to 20 and 196 n_p is equal to 12 after finite difference analysis first screening. This lead to 261 197 computations.

2.6. Stochastic analysis

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The uncertainty quantification is performed using Monte-Carlo sampling me-200 thod. For the uncertainty quantification, 750 modal bases are computed. The comparison of the matched eigenmodes between nominal model results and the ones obtained with modified input parameters is performed. Material and climatic parameters are defined as equiprobabilistic distributions between upper and lower values given in the table 1. The matched eigenfrequencies are given as box 205

and whiskers plots and the experimental data are compared with the computed distributions. The error between numerical eigenfrequencies and averaged measured eigenfrequencies is compared with the relative standard deviation (RSD) of each numerical eigenfrequencies.

210 2.7. CFDAC

A frequency domain assurance criterion is used, based on the experimental and computed FRFs. This method can be regarded as the equivalent of the Modal Assurance Criterion in the FRF domain. The FDAC [28], adapted as Complex-FDAC, is expressed in the eq. 6 [29]:

$$CFDAC_{fg} = \frac{\left[\sum_{i=1}^{N} \sum_{j=1}^{N} h_{ij}(\omega_f) h_{ij}^{d}(\omega_g)\right]^2}{\left[\sum_{i=1}^{N} \sum_{j=1}^{N} h_{ij}(\omega_f) h_{ij}(\omega_f)\right] \left[\sum_{i=1}^{N} \sum_{j=1}^{N} h_{ij}^{d}(\omega_g) h_{ij}^{d}(\omega_g)\right]}$$
(6)

In this equation, i and j correspond to the excitation and measure FRF, h_{ij} refers to pristine state FRFs, h_{ij}^d refers to altered FRFs. f and g refer to each pair of spectral lines compared from the two sets of mobility functions. N is the number of sampled points in the specimen. CFDAC results in a complex two-dimensional array of dimension N×N. Real and imaginary parts of the CFDAC are absolute-valued even when used for numeric computations. The CFDAC is performed between real soundboards and model nominal cases to observe the discrepancies in the behaviour of the soundboards.

2.23 2.8. Fuzzy-FRF analysis

The Fuzzy frequency response function (FUZZY-FRF) is computed using the 750 FRF that are computed during the uncertainty quantification process. For each case, admittance is computed. The FRF used are located at the same point than experimental ones. The 750 FRF are gathered in the same plot, and space of the plot is discretised in 2500×2500 subspaces. In each subspace, the number of lines passing through the subspace is evaluated. A color scale is applied to represent the amount of lines per subspace. This method is taken from [30, 31].

231 2.9. Modal overlap factor

The modal overlap factors (MOF) by third octave bands are calculated using the eq 7

$$MOF = M_d \times \eta \times F_c \tag{7}$$

With M_d the modal density, given by the ratio between the number of modes per third octave bands, for a central frequency F_c . η is the loss factor of the system in the considered bandwidth and is equal to 2.3 %, two times the average value of the modal damping, given in the table 3. Only one domain is considered, the low-frequencies where the MOF value is comprised between 0 and 30 %.

239 3. Results

In this section the experimental and numerical results are given in different subsections.

242 3.1. Experimental and numerical deformed shapes

The measures on the five soundboards have led to five experimental modal 243 bases. The values of the first eight eigenmodes frequencies and corresponding 244 modal dampings are given in the table 3. The first eight numerical eigenmodes 245 shapes and nominal eigenfrequencies are given in the figure 4. The eigenmodes shape consist in torsion modes in the LR plane, flexure modes in the L and R directions of the soundboard and mixed torsion and flexure modes. The mo-248 dal overlap factor values are given in the table 2. Below 500 Hz, the modal 249 overlap factor is comprised between 2 and 28 % which corresponds to the low 250 frequency domain, where, mainly, the modal analysis is relevant. Above 500 Hz 251 the mid-frequencies domain is reached, which suggests an increase of discrepancies between experimental and numerical results. 253

3.2. Co-located nominal FRF comparison

The co-located FRF in acceleration of the nominal model and the experimental soundboards are given in the figure 3. It is shown that the level of the dynamical responses of the experimental soundboards are close to the computed one up, and it is not possible to differentiate them based on the acceleration level.

3.3. Comparison between numerical and experimental eigenfrequencies, deterministic approach

The eigenfrequencies, as well as their mean and standard deviation, for the first eight modes are given in the table 3. The average relative standard deviation (RSD) of the experimental eigenfrequencies is equal to 4.3 % and the average modal damping is equal to 1.15 % on the modes considered. The relative error between nominal model eigenfrequencies and experimental ones is comprised between -6.3 and 7.2 %.

268 3.4. Spectral correlation of numerical and experimental guitar soundboard

The figure 5 shows the frequency domain assurance criterion from 0 to 800 Hz for each case. For the first four sboundboard, up to 600 Hz, the CFDACs show a good correlation between experimental and numerical databases, which corresponds to the domain of low-frequencies, as shown in the table 2. Moreover, the CFDAC matrix complex correlation is almost diagonal, which indicates a good correspondence between stiffness and mass of the model and the real soundboards.

276 3.5. Stochastic analysis results

The results of the uncertainty quantification study are given for the first eight modes as box and whiskers plots on the figure 6 and mean, SD and RSD in the table 4. In the figure 6, the in boxes vertical lines correspond to the median, the lower and upper limits of the boxes correspond to the lower and upper first quartile (25 percentiles) respectively and the limits of the left and right

whiskers correspond to 9 and 91 percentiles respectively. It is shown that all the experimental eigenfrequencies are comprised between the 9 and 91 percentiles.

Moreover, most of the experimental eigenfrequencies are comprised in the second quartile (50 percentiles).

In the table 4, it is shown that the RSD of the first eight numerical eigenfre-

quencies is rather diffuse and comprised between 5.8 and 8.2 %, with an average value equal to 6.9 %.

289 3.6. Fuzzy-FRF of a free edges soundboard of Spanish guitar soundboard

The figure 7 (a) shows the Fuzzy-FRF evaluated with the database used 290 for the uncertainty quantification, which corresponds to the co-located FRF 291 in displacement of the 750 computations. This figure gives a display of the variability distribution of the FRF of the studied structures when undergoing 293 material and climatic variations. The figure 7 (a) displays the number of FRF 294 that are comprised in each discretised subspace of the plot. Each axis is divided 295 2500 times, which means that the figure displays 6.25×10^6 discretised squares. 296 So, the sampling is equal to 0.32 Hz for the frequency axis and from 5.10^{-3} to $40~m.s^{-2}.N^{-1}$ for the acceleration axis. The maximum number of FRF inside 298 a square is equal to 100, at the lowest frequency where the dispersion is the 299 lowest. This value is used as the maximum value for the color fringe and the 300 lowest value 0 corresponds to the case where no FRF crosses a discretised square. 301 The dispersion of the FRF increases above 100 Hz, and becomes rather diffuse above this value. The figure 7 (b) gives the min and max values of the FRF, 303 represented as dashed black lines. The experimental FRF are also displayed 304 in the figure 7 (b), and are generally comprised inside the area of presence 305 probability, where the number of occurrences is higher than 30. 306

3.7. Screening analyses results

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The ranking of the elementary effects of the material and climatic parameters is given for each modes in the figure 4. Depending on the eigenmode shape, the influential material parameters ranking varies, and is in accordance with the

finite differences sensitivity matrix given in the figure 8 (a). As a primary result of the sensitivity matrix, it is shown that the specific elastic parameters of the 312 soundboard are the most influential. The specific elastic parameters, when their 313 complete range is considered are more influential on the eigenfrequencies than 314 the densities. The relative humidity impact on the eigenfrequencies is almost 315 constant for each mode and corresponds to values comprised between 14 and 18 316 % of the total of the elementary effects. The ranking of the material and climatic parameters for global dynamical be-318 haviour of the first 20 modes is given in the figure 8 (a) for the eigenfrequencies 319 and (b) for the eigenvectors. It is shown that the influence of each parameter 320 varies according to the dynamical feature observed. Generally, in free conditions, 321 the specific rigidity in the longitudinal direction of the soundboard plate is the 322 most influential parameter (22 % of the total), followed by the specific shear rigidity in LR plane (19 %) and the specific radial rigidity (15 %). The impact 324 of the density of the soundboard is similar to the impact of the density of the 325 braces (9 and 8 %, respectively). The remaining specific elastic parameters of 326 the braces are less influential, and correspond to 7 % for the braces L rigidity 327 and 3 % for the LR shear elasticity of the braces. The RH has a strong impact 328 on the eigenfrequencies, corresponding to 17 % of the total elementary effects. 329 The effect of the RH is smaller on the eigenvectors; as its effect is global on 330 every elastic and density parameter, it affects in a smaller way the eigenmode 331 shapes. In has to be pointed out that, considering eigenvectors, the longitudinal specific modulus of the soundboard is mainly influential, up to two times the 333 specific rigidities in R direction or LR plane. 334

4. Discussion 335

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The results obtained lead to multiple discussions. First of all, the determi-336 nistic comparison has highlighted the good predictive capability of the model concerning the dynamical behaviour of Spanish guitar soundboards. It has been shown that the relative error between a nominal model (whose material and 339

climatic parameters are taken form the literature) and the average values of 340 eigenfrequencies is close to 4\%, which is close to the relative standard deviation 343 of the experimental eigenfrequencies. When material and climatic uncertainties are taken into account, the model is able to predict every experimental frequen-343 cies, which are, in addition, comprised in the second quartile (50 percentiles) 344 of the computed ones. The comparison of deterministic numerical and experi-345 mental FRF have shown that the discrepancies increase above 500 Hz which is close to the limit of the low frequencies domain given by the modal overlap factor. This result is in accordance with dynamics theory, and highlights the 348 limits of a modal point of view for this type of study. The CFDAC compari-349 son between experimental and numerical FRF is innovative in this application 350 domain, enables a global comparison and is more adapted in mid-frequencies 35 range. The CFDAC correlation criterion is a relevant quantified global indicator of closeness between numerical and experimental data. The Fuzzy-FRF feature 353 proposed here shows the dispersion of the model response for a fixed design. 354 Thus, a guitar soundboard built with the same geometry and the same wood 355 species will exhibit a dynamical behaviour that may be included in the Fuzz-356 FRF prediction. As shown in this study, experimental FRF were comprised in 357 the fuzzy-FRF high plausibility area and inside the upper and lower bounds of 358 the stochastic simulations, as shown in the figure 7 (b). This post-processing of 359 the FRF in the case of an uncertainty quantification can be a useful tool for the 360 decision-support in musical instruments making, and to confirm the relevance of a design change over the material and climatic variability impact. Based on the fact that the model was able to correctly predict dynamical beha-363 viour, it has been shown that the most influential parameters with regard to the 364 eigenfrequencies and eigenvectors of a free conditions soundboard were mainly 365 specific elastic parameters $\frac{E_L}{\rho}$, $\frac{E_R}{\rho}$ and $\frac{G_{LR}}{\rho}$ of the top plate soundboard, followed by the relative humidity and the density of the plate and braces. These results are similar to the one obtained considering a violin in [32]. It has been 368 shown in [33] and [21] that studies on tonewood were most of the time focused on $\frac{E_L}{\rho}$, which is also correlated with high "quality" wood. Considering these new results, it is clear that, as $\frac{E_R}{\rho}$ and $\frac{G_{LR}}{\rho}$ play an important role, the selection criteria and studies should also focus on these parameters.

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374 Conclusion

In this paper, the comparison between experimental and numerical dynami-375 cal data have shown the capability of physic-based models to predict complex 376 assemblies responses. It has been shown that model accuracy was maintained 377 even in the presence of strong material and climatic uncertainties. This paper 378 proposes new ways for the characterisation of musical instruments: the uncertainty quantification for a given geometry, the CFDAC between model and 380 real instruments and the Fuzzy-FRF for the post processing of the study of the 381 uncertainties effects on the dynamical behaviour of musical instruments. This 382 is a first and innovative step in the validation process of physics-based models 383 of musical instruments, which, associated with the different dynamical features proposed for this domain can be used for design and restoration purposes of musical instruments. Moreover, the results obtained have questioned the cur-386 rent selection criterion for spruce tonewood, and shown that specific moduli 387 other than $\frac{E_L}{\rho}$ were also significant on the dynamics of the guitar soundboard in free conditions, which is its main boundary conditions during making steps. 389 These results are relevant in the low frequency domain, where modal analysis is 390 effective. In order to manage with higher domains, such as mid-frequencies and 391 high frequencies domains, large frequency bandwidth descriptors [34], such as 392 mean-values approaches should be considered, and have already proved useful for acoustic guitars [13], violins [35] and composite plates [36]. Modal approaches 394 and medium and high frequencies methods would provide, through model vali-39 dation process, reliable large band datasets of musical instruments behaviours, 396 even in presence of strong uncertainties.

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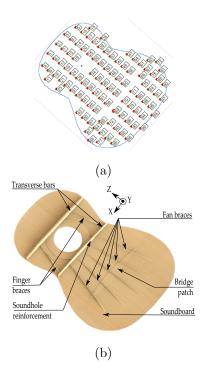
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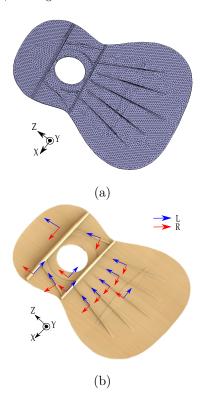
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FIGURE 1: (a) location of the experimental measurement points and FRF synthesis nodes used in [19] experiments; (b), Computer Aided Design and nomenclature of Spanish guitar soundboard.



 $\label{eq:figure 2} Figure \ 2: \ (a) \ finite \ element \ model \ of \ the \ guitar \ soundboard \ ; \ (b) \ orientation \ of \ the \ material, \\ L: \ longitudinal, \ R: \ radial, \ T: \ tangential.$



 $\label{eq:Figure 3: Inertances of the colocated measured and excitation point for both model and experimental FRF.$

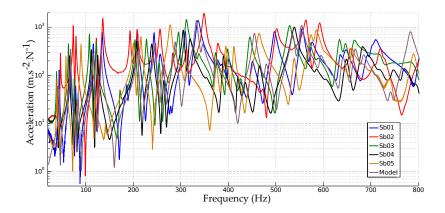


Table 1: Material properties of spruce implemented in the numerical model; italic from [21] at MC=10%. Remaining values from [37] and [38]. Loss factors values are given but have not been implemented for undamped modal analysis.

Parameter	Mean value	Min. value	Max. value
$\frac{E_L}{\rho} (\text{MPag}^{-1} \text{cm}^{-3})$	29000	20590	35380
$\frac{\dot{E}_R}{\rho}$ (MPag ⁻¹ cm ⁻³)	2280	1460	3810
$\frac{\dot{E}_T}{\rho}$ (MPag ⁻¹ cm ⁻³)	1480	1300	1660
$ u_{LR}$ (-)	0.37	-	-
$ u_{RT}$ (-)	0.48	-	-
$ u_{TL}$ (-)	0.02	-	-
η_L (%)	0.73	0.09	0.12
η_R (%)	1.7	0.17	0.1
η_{LR} (%)	1.2	0.2	0.17
$\frac{G_{LR}}{\rho}$ (MPag ⁻¹ cm ⁻³)	1850	1295	2442
$\frac{G_{RT}}{\rho}$ (MPag ⁻¹ cm ⁻³)	100	74	150
$\frac{G_{TL}}{\rho}$ (MPag ⁻¹ cm ⁻³)	1910	1070	2750
Density $(g cm^{-3})$	0.44	0.39	0.51
Relative humidity (%)	50	20	85
Temperature ($^{\circ}$ C)	21	15	35

Table 2: Values of modal overlap factor for corresponding third octaves bands.

Third octave band	M.O.F. (%)	Domain
40	8	L.F.
50	2	L.F.
63	2	L.F.
80	14	L.F.
100	4	L.F.
125	10	L.F.
160	7	L.F.
200	9	L.F.
250	18	L.F.
315	28	L.F.
400	22	L.F.
500	34	M.F.
630	52	M.F.

FIGURE 4: Left side, frequencies and deformed shapes of the first eight computed modes of the spanish guitar soundboard, in free-free conditions. The red color represents the highest eigenvectors values. Right side, elementary effect ranking for each parameter using Morris sensitivity Method.

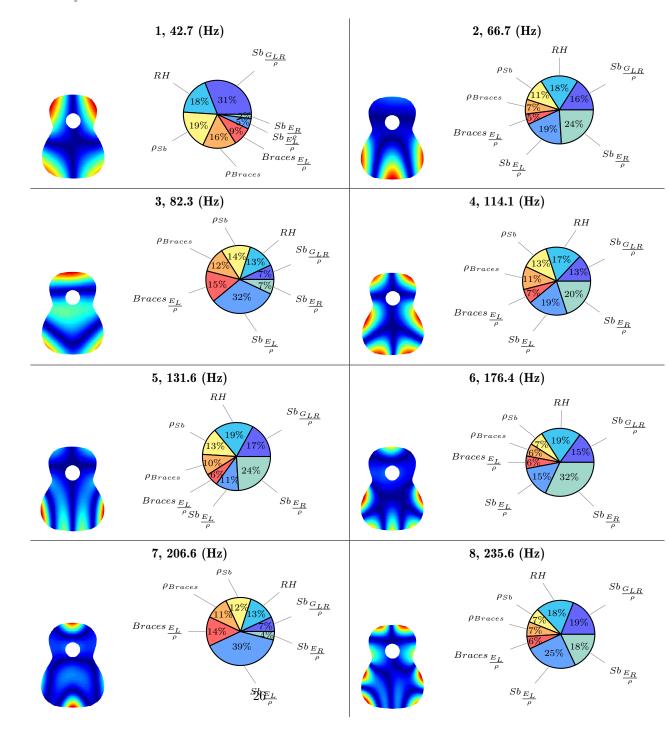


Table 3: Results for the first eight modes measured and computed. Mean, standard deviation (SD) and relative standard deviation (RSD) of the measured eigenfrequencies, as well as the average modal damping are given.

Mode	Num. (Hz)	Sb_{01} (Hz)	Sb_{02} (Hz)	Sb_{03} (Hz)	Sb_{04} (Hz)	Sb_{05} (Hz)	Av. exp (Hz)	SD. exp (Hz)	RSD. exp	Num. vs Exp. error (%)	Average ξ (%)
1	42.7	42.4	45.2	37	38.2	37.8	40.1	2.9	7.3	6.5	0.6
2	66.7	66.3	72.3	62	64.5	56.9	64.4	4.0	6.1	3.6	0.7
3	82.3	77.2	77.8	74.4	77	75.8	76.4	1.1	1.4	7.2	0.7
4	114.1	114	120.9	106.2	110.5	117	113.7	4.3	3.8	0.3	1.1
5	131.6	133	134	121	123	140	130.2	6.6	5.0	1.1	1.2
6	176.4	185	192	170	-	199	186.5	9.0	4.8	-5.7	1.3
7	206.6	208	207	202	-	214	207.8	3.3	1.6	-0.7	1.7
8	235.6	242	245	253	237	275	250.4	10.9	4.3	-6.3	2.0
Mean	-	-	-	-	-	-	-	5.2	4.3	3.9 (abs.values)	1.15

Table 4: Mean, SD and RSD of the matched numerical eigenfrequencies.

Mode	Mean (Hz)	SD (Hz)	RSD (%)
1	42	2.8	6.7
2	66.6	4.9	7.3
3	80.7	4.7	5.8
4	112.8	7.7	6.8
5	133.1	10.2	7.7
6	177.1	14.6	8.2
7	201.4	11.8	5.9
8	233.6	6.3	7.0
Mean	-	-	6.9

FIGURE 5: CFDAC matrix (magnitude) between the measured (ordonate) and computed with the nominal model (abscissa) FRF for each soundboard.

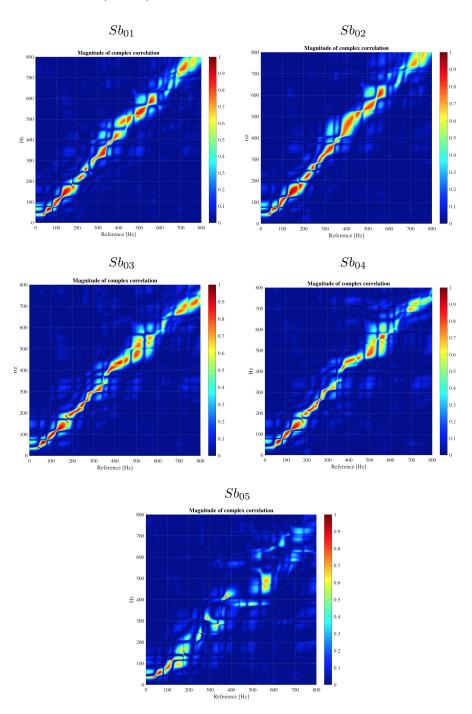


FIGURE 6: Box and whiskers plots of the values of the first eight matched eigenfrequencies computed by the numerical model of soundboard, comparison with experimental eigenfrequencies values. The vertical lines of the boxes correspond, from left to right, to the first quartile (25^{th} percentiles) the median and the third quartile. The limits of the left and right whiskers correspond to the 9^{th} and 91^{th} percentiles, respectively.

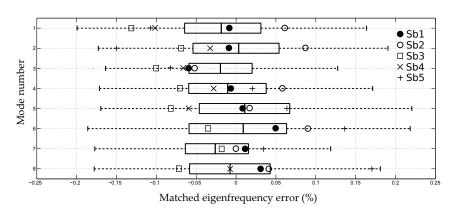


FIGURE 7: (a) fuzzy-FRF of the co-located admittances from the uncertainty quantification computations, the total number of runs is equal to 750; (b) statistical treatment of the FRFs, dashed black line, upper and lower limits of the stochastic computations.

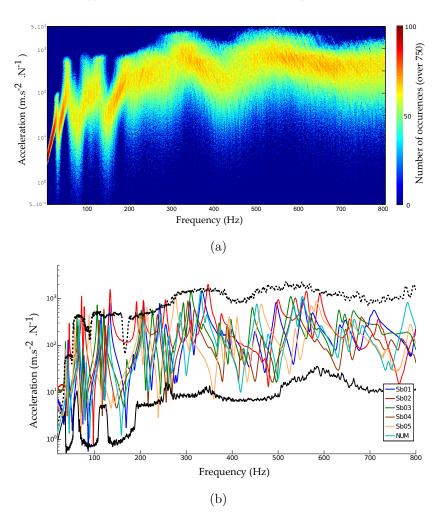


FIGURE 8: Finite difference sensitivity matrix for the first 30 modes (a); relative elementary effects of the material and climatic parameters in regard with eigenfrequencies (b) and eigenvectors (c).

